

Dynamic response of graphene-platelets reinforced circular plates subjected to low-velocity impact with spinning motion

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Abstract. This study pioneers the analysis of low-velocity impact behavior in spinning functionally graded porous circular plates reinforced with graphene platelets (GPLs), employing first-order shear deformation theory. The governing equations are derived via Hamilton's principle, while the plate-impactor interaction is modeled using a modified Hertzian contact law. Numerical solutions are validated against existing literature showing excellent agreement. Parametric studies investigate the effects of: (1) Spinning speed (revealing that higher speeds increase contact force but reduce impact displacement), (2) Porosity characteristics (distribution patterns and coefficients), (3) GPL reinforcement configurations, and (4) Nanofiller weight fractions on the dynamic response. Results highlight that porosity distribution has the most pronounced influence on both contact force evolution and transient center-point deflection, outweighing other variables.

Keywords: circular plate; graphene platelets; low-velocity impact; metal foam; spinning motion

1. Introduction

Low-velocity impact is critically important in engineering safety. It involves the damage initiation, propagation, and failure mechanisms of materials/structures under low-speed collisions, directly affecting safety in key sectors like aerospace and transportation (Song *et al.* 2019, Yang *et al.* 2022, Fan and She 2026, Fan *et al.* 2025a, 2025b). Statistics indicate approximately 15% of mechanism-related accidents stem from cumulative low-velocity impact damage. Research in this area optimizes impact-resistant designs, extends equipment lifespan, and reduces maintenance costs. Current studies focus on predicting the dynamic response of composite materials and innovating protective structures, holding profound implications for ensuring personnel safety and structural integrity. Furthermore, research articles specifically on low-velocity impact on circular plates are very limited, particularly within this existing literature. This overview summarizes key research on the low-velocity impact (LVI) behavior of composite circular plates: Khatounabadi *et al.* (2025) utilized a modified Hertzian contact law and finite element method (FEM) to analyze LVI on porous circular plates reinforced with graphene platelets. Raissi (2021) investigated the

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dynamic response of composite plates impacted by a cylindrical striker with an initial velocity of 1 m/s. Mars *et al.* (2018) pioneered the LVI analysis of elastoplastic functionally graded material (FGM) plates. Their approach integrated the Mori-Tanaka model for micromechanical property determination and Suquet's self-consistent method for stress-field homogenization. Gunes *et al.* (2014) explored the LVI behavior of clamped functionally graded circular plates through combined experimental testing and explicit finite element modeling. Shariyat and Jafari (2013a) developed a weighted residual semi-analytical solution with an iterative scheme to analyze LVI on functionally graded plates, highlighting limitations of the simple rule of mixtures. Dai *et al.* (2013) studied the nonlinear dynamic response of functionally graded circular plates subjected to LVI, employing Giannakopoulos' contact model to derive expressions for contact force and central deflection. Shariyat and Jafari (2013b) presented a Galerkin-type solution procedure for the nonlinear LVI analysis of circular plates with bidirectional material gradation under radial preloads. Gunes *et al.* (2011) conducted a three-dimensional elastoplastic analysis of functionally graded (SiC/Al) circular plates experiencing low-velocity impact loads.

While rotational motion is frequently neglected in contemporary circular plate vibration studies, its influence necessitates consideration. Recent research addressing this gap includes: Mirdrikvandi *et al.* (2025) applied the differential quadrature method to examine free vibration in rotating sandwich annular plates featuring porous cores and carbon nanotube-reinforced polymeric nanocomposite face sheets. Morozov *et al.* (2024) developed a model for the coupled plane-transverse vibrations of rotating thin circular plates with concentric holes, explicitly incorporating Coriolis and centrifugal inertial forces generated by in-plane rotation. Li and Hu (2021) explored principal and 3:1 internal resonances in rotating, edge-clamped conductive circular plates within an air-magnetic field, identifying period-doubling bifurcations in intermittent periodic responses. Hu and Li (2019) investigated magnetoelastic axisymmetric multi-mode interactions and Hopf bifurcations in conductive circular plates rotating through air and transverse magnetic fields using multiple scales combined with polar coordinate transformation. Yang *et al.* (2019) studied the thermoelastic coupling effects on vibration and stability in rotating friction circular plates, incorporating middle internal forces and deformation effects. Hu and Li (2018) characterized primary resonance and bifurcation phenomena in rotating conductive plates subjected to air-magnetic fields, employing bifurcation diagrams, maximum Lyapunov exponents, and parametric response analysis. Hu *et al.* (2018) examined nonlinear magneto-elastic combined resonances (parametric and forced) in circular plates rotating with variable speed under alternating load, based on derived magneto-elastic vibration equations. Hu and Wang (2016) focused on the nonlinear free vibration of conductive rotating thin circular plates under static loads within a magnetic field, deriving the governing nonlinear equations of motion. Hu and Wang (2015) utilized a modified multiple scale method to determine amplitude-frequency characteristics and relationships between amplitude, static loads, and excitation forces for the plate. Younesian *et al.* (2015) presented analytical closed-form solutions (series expansions) for deflection, strain, and stress distributions in hollow circular plates subjected to a rotating peripheral force. Li *et al.* (2011) derived dynamic governing equations and boundary conditions for rotating laminated circular plates with viscoelastic core layers using Hamilton's principle. Bui *et al.* (2024) investigated the thermo-mechanical dynamic buckling and vibration of FG-GPLRC circular plates and shallow spherical shells on nonlinear viscoelastic foundations. Nam *et al.* (2025) developed a semi-analytical method for analyzing the nonlinear dynamic responses of functionally graded porous graphene platelet-reinforced circular plates and spherical shells. Nguyen *et al.* (2024) examined the thermo-mechanical axisymmetric stability of FG-GPLRC spherical shells and circular plates on nonlinear

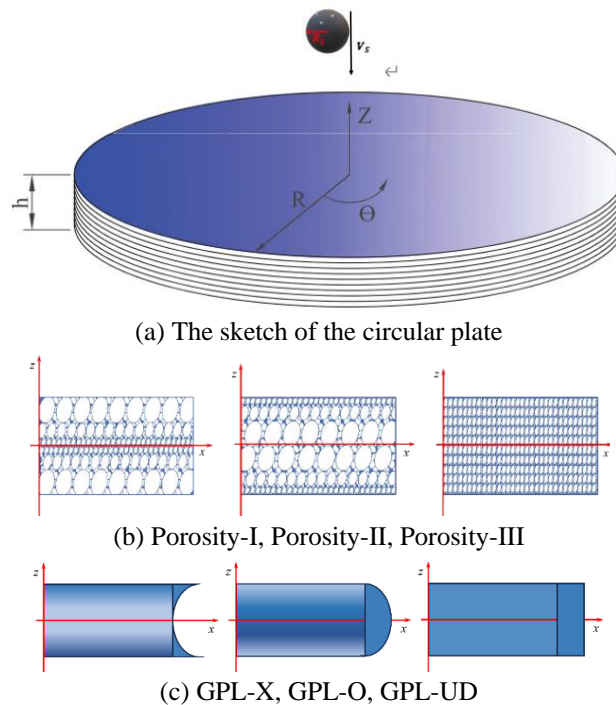


Fig. 1 The sketch of low-velocity impact response of graphene-reinforced porous circular plates

elastic media. Meanwhile, Vu *et al.* (2025a) studied the vibration responses of graphene platelet-reinforced circular plates and spherical shells with multi-step spiderweb stiffeners and piezoelectric layers. In another work, Vu *et al.* (2025b) proposed a semi-analytical approach for analyzing the nonlinear electro-thermo-mechanical dynamic responses of FG-GPLRC shallow spherical caps and circular plates with porous cores.

The preceding literature review reveals that existing research on the dynamic analysis of rotating circular plates focuses primarily on resonant vibrations (free and forced), with relatively few dedicated studies (Hu and Ma 2019, Li *et al.* 2017). Crucially, no prior investigation addresses the low-velocity response of these structures subjected to low-velocity impact loads considering the effect of spinning motion. To address this gap, this study pioneers the exploration of this specific low-velocity regime. The objectives of this paper is to develop a mathematical model for the rotating circular plate under low-velocity impact loads, incorporating three distinct material distribution patterns and accounting for rotational dynamics, and implement a computational framework based on the finite difference method combined with an iterative technique to solve the resulting nonlinear low-velocity dynamic equations, and conduct numerical simulations to demonstrate the influence of key parameters on the impact force history and the plate's temporal response.

2. Theoretical modeling

Fig. 1 shows the circular plate, having thickness h and radius R . The displacement components

u , v , and w correspond to the mid-plane directions r , θ , and z within the cylindrical coordinate system centered there. Three porosity distributions across the plate's thickness are defined in Fig. 2. This study considers three distinct porosity distributions through the thickness (z -axis) of the circular plate: a uniform distribution (Porosity-III) and two non-uniform symmetric distributions (Porosity-I and Porosity-II). The corresponding variations in material properties along the z -axis are as follows (Zhou *et al.* 2024, Kalhori *et al.* 2023a, 2023b, Caglar *et al.* 2025, Shen *et al.* 2024, Wang *et al.* 2023, Wu *et al.* 2023, Xia *et al.* 2023, Zhao *et al.* 2024a, 2024b, Cheng *et al.* 2025, Cheng and She 2025, Ma *et al.* 2025, She and He 2025),

For Porosity-I

$$\begin{aligned} E(z) &= E^* \left[1 - e_1 \left\{ 1 - \cos\left(\frac{\pi z}{h}\right) \right\} \right], \\ \rho(z) &= \rho^* \left[1 - e_{m1} \left\{ 1 - \cos\left(\frac{\pi z}{h}\right) \right\} \right], \\ \mu(z) &= \mu^*. \end{aligned} \quad (1)$$

For Porosity-II

$$\begin{aligned} E(z) &= E^* \left[1 - e_2 \cos\left(\frac{\pi z}{h}\right) \right], \\ \rho(z) &= \rho^* \left[1 - e_{m2} \cos\left(\frac{\pi z}{h}\right) \right], \\ \mu(z) &= \mu^*. \end{aligned} \quad (2)$$

For Porosity-III

$$E(z) = E^* e_3, \quad \rho(z) = \rho^* e_{m3}, \quad \mu(z) = \mu^*. \quad (3)$$

Herein, the plate's position-dependent material properties are defined as: elastic modulus $E(z)$, mass density $\rho(z)$, Poisson's ratio $\mu(z)$; Mass density coefficients are denoted e_{m1} , e_{m2} , e_{m3} ; while porosity size parameters are e_1 , e_2 , e_3 ; and the relationship between e_{mi} and e_i are as follows (Zhang *et al.* 2024)

$$\begin{cases} 1 - e_{m1} \left\{ 1 - \cos\left(\frac{\pi z}{h}\right) \right\} = \sqrt{1 - e_1 \left\{ 1 - \cos\left(\frac{\pi z}{h}\right) \right\}}, & \text{(GPL-X)} \\ 1 - e_{m2} \cos\left(\frac{\pi z}{h}\right) = \sqrt{1 - e_2 \cos\left(\frac{\pi z}{h}\right)}, & \text{(GPL-O)} \\ e_{m3} = \sqrt{e_3}, & \text{(GPL-UD)} \end{cases} \quad (4)$$

To comparatively assess various porosity and GPL distribution patterns, all plate configurations are constrained to maintain identical total mass across different porosity-GPL combinations. This yields (Nam *et al.* 2025)

$$\int_0^{\frac{h}{2}} \sqrt{1 - e_1 \left\{ 1 - \cos\left(\frac{\pi z}{h}\right) \right\}} dz = \int_0^{\frac{h}{2}} \sqrt{1 - e_2 \cos\left(\frac{\pi z}{h}\right)} dz = \int_0^{\frac{h}{2}} \sqrt{e_3} dz \quad (5)$$

Where e_1 (a given input) defines the porosity coefficients e_2 and e_3 . And, these non-porous material properties ($E^*, \mu^*, \rho^*, \alpha^*$) have the following form (Jafary *et al.* 2025)

$$E^* = E_m \left(\frac{3}{8} \left(\frac{1 + \zeta_L \eta_L V_{GPL}}{1 - \eta_L V_L} \right) + \frac{5}{8} \left(\frac{1 + \zeta_W \eta_W V_{GPL}}{1 - \eta_L V_{GPL}} \right) \right), \tag{6}$$

$$\begin{aligned} \rho^* &= \rho_{GPL} V_{GPL} + \rho_m V_m, \\ \mu^* &= \mu_{GPL} V_{GPL} + \mu_m V_m. \end{aligned} \tag{7}$$

Herein, E_{GPL} and E_m stand for Young’s modulus of GPLs and matrix. The same subscript notation applies to Poisson’s ratio ν_{GPL} (ν_m), and mass density ρ_m (ρ_{GPL}), and $\zeta_L = \frac{2L_{GPL}}{t_{GPL}}$, $\zeta_W = \frac{2W_{GPL}}{t_{GPL}}$, $\eta_L = \frac{E_{GPL} - E_m}{E_{GPL} + \zeta_L E_m}$, $\eta_W = \frac{E_{GPL} - E_m}{E_{GPL} + \zeta_W E_m}$.

For a transversely deforming circular plate under FSDT, displacements primarily depend on radial coordinate r , angular coordinate θ , and time t . The radial (u_r) and circumferential (u_θ) components respectively describe lengthwise and tangential position changes. The displacement fields take the form (Zhang and Zhou 2025, Abo *et al.* 2025, Lahdiri *et al.* 2024, Kumar *et al.* 2024, Zohra *et al.* 2024, Zhang *et al.* 2023, Li *et al.* 2024)

$$\begin{aligned} u(r, \theta, z, t) &= u_0(r, \theta, t) + z\varphi_r(r, \theta, t), \\ v(r, \theta, z, t) &= v_0(r, \theta, t) + z\varphi_\theta(r, \theta, t), \\ w(r, \theta, t) &= w_0(r, \theta, t) \end{aligned} \tag{8}$$

The midplane displacements along r , θ , and z axes are designated u_0 , v_0 , and w_0 respectively. Rotational components are denoted ϕ_r (circumferential plane) and ϕ_θ (transverse plane). Under FSDT, selection of an appropriate shear correction factor K_s is critical. However, determination of K_s for composite circular plates remains complex. Thus, we adopt $K_s = \pi^2/12$ throughout this study for simplification (Abo *et al.* 2025, Lahdiri *et al.* 2024, Kumar *et al.* 2024, Zohra *et al.* 2024). Eq. (8) defines the stress-strain relationship as

$$\begin{aligned} \varepsilon_{rr} &= \varepsilon_{rr}^0 + z\varepsilon_{rr}^1, \quad \varepsilon_{\theta\theta} = \varepsilon_{\theta\theta}^0 + z\varepsilon_{\theta\theta}^1, \\ \varsigma_{r\theta} &= \varsigma_{r\theta}^0 + z\varsigma_{r\theta}^1, \quad \varsigma_{rz} = \varsigma_{rz}^0, \quad \varsigma_{\theta z} = \varsigma_{\theta z}^0 \end{aligned} \tag{9}$$

with

$$\begin{aligned} \left\{ \begin{matrix} \varepsilon_{rz}^0 \\ \varepsilon_{\theta z}^0 \end{matrix} \right\} &= \left\{ \begin{matrix} \frac{\partial w_0}{\partial r} + \varphi_r \\ \frac{1}{r} \frac{\partial w_0}{\partial \theta} + \varphi_\theta \end{matrix} \right\}, \quad \left\{ \begin{matrix} \varepsilon_r^1 \\ \varepsilon_\theta^1 \\ \varsigma_{r\theta}^1 \end{matrix} \right\} = \left\{ \begin{matrix} \frac{\partial \varphi_r}{\partial r} \\ \frac{\partial \varphi_\theta}{\partial \theta} \frac{1}{r} + \frac{1}{r} \varphi_r \\ \frac{\partial \varphi_r}{\partial \theta} \frac{1}{r} + \frac{\partial \varphi_\theta}{\partial r} \end{matrix} \right\}, \\ \left\{ \begin{matrix} \varepsilon_{rr}^0 \\ \varepsilon_{\theta\theta}^0 \\ \varsigma_{r\theta}^0 \end{matrix} \right\} &= \left\{ \begin{matrix} \frac{\partial u_0}{\partial r} + \frac{1}{2} \left(\frac{\partial w_0}{\partial r} \right)^2 \\ \frac{\partial v_0}{\partial \theta} \frac{1}{r} + \frac{1}{2r^2} \left(\frac{\partial w_0}{\partial \theta} \right)^2 + \frac{u_0}{r} \\ \frac{\partial v_0}{\partial r} + \frac{\partial u_0}{\partial \theta} \frac{1}{r} + \frac{\partial w_0}{\partial r} \frac{\partial w_0}{\partial \theta} \frac{1}{r} \end{matrix} \right\}. \end{aligned} \tag{10}$$

The constitutive relation can be described as

$$\begin{bmatrix} \sigma_{rr} \\ \sigma_{\theta\theta} \\ \sigma_{rz} \\ \sigma_{\theta z} \\ \sigma_{r\theta} \end{bmatrix} = \begin{bmatrix} Q_{11} & Q_{12} & 0 & 0 & 0 \\ Q_{12} & Q_{22} & 0 & 0 & 0 \\ 0 & 0 & Q_{44} & 0 & 0 \\ 0 & 0 & 0 & Q_{55} & 0 \\ 0 & 0 & 0 & 0 & Q_{66} \end{bmatrix} \cdot \begin{bmatrix} \varepsilon_{rr} \\ \varepsilon_{\theta\theta} \\ \zeta_{rz} \\ \zeta_{\theta z} \\ \zeta_{r\theta} \end{bmatrix} \quad (11)$$

The stiffness coefficients are $Q_{11} = Q_{22} = \frac{E(z)}{1-\mu(z)^2}$, $Q_{12} = \frac{\mu(z)E(z)}{1-\mu(z)^2}$, $Q_{44} = \frac{E(z)}{2[1+\mu(z)]}$, $Q_{55} = Q_{66} = K_s Q_{44}$.

Hamilton’s principle provides the circular plate governing equations (Hu and Xu 2021)

$$\int_t (-\delta U + \delta K - \delta U_r + \delta U_f + \delta W) dt = 0 \quad (12)$$

The energy terms comprise: U (strain energy), K (kinetic energy), U_r (centrifugally induced potential), U_f (damping virtual strain) and W (impact load work).

$$U_r = \frac{1}{2} \int_{R_b}^{R_a} \int_0^{2\pi} \int_{-h/2}^{h/2} -\rho(z)\Omega^2 urdzd\theta dr, \quad (13)$$

$$W = - \int_{R_b}^{R_a} \int_0^{2\pi} [F_i(t)\delta(w_s - w)w - c_t \dot{w}] rd\theta dr, \quad (14)$$

$$K = \int_{R_b}^{R_a} \int_0^{2\pi} \int_{-h/2}^{h/2} \frac{1}{2} \rho(z) \left[\begin{matrix} (\frac{\partial u}{\partial t} - \Omega v)^2 \\ + [\frac{\partial v}{\partial t} + \Omega(r + u)]^2 \\ + (\frac{\partial w}{\partial t})^2 \end{matrix} \right] rdz d\theta dr, \quad (15)$$

$$U = \frac{1}{2} \int_{R_b}^{R_a} \int_0^{2\pi} \int_{-h/2}^{h/2} (\sigma_{rr}\varepsilon_{rr} + \sigma_{\theta\theta}\varepsilon_{\theta\theta} + \sigma_{rz}\zeta_{rz} + K_s\sigma_{\theta z}\zeta_{\theta z} + K_s\sigma_{r\theta}\zeta_{r\theta}) rdz d\theta dr. \quad (16)$$

Here, Ω is the rotating speed, ct is the damping coefficient, $F_i(t)$ denotes the Hertzian contact force, with separate loading/unloading models per (Gunes *et al.* 2014, 2011)

$$F_i(t) = \begin{cases} \left[\frac{4}{3} \left(\frac{1-\mu_s^2}{E_s} + \frac{1}{E_{11}} \right)^{-1} \sqrt{R_s} \right] (w_i - w_s)^{3/2}, (\text{loading}) \\ F_{\max} \left(\frac{\alpha}{\alpha_{\max}} \right)^{5/2}, (\text{unloading}) \end{cases} \quad (17)$$

Definitions: E_{11} : Elasticity modulus (plate’ surface), E_s , μ_s : Impactor elasticity modulus and Poisson’s ratio, α_{\max} : Maximum indentation depth, F_{\max} : Maximum impact force, and Ω symbolize rotating velocity. Inserting Eqs. (13)-(16) into Eq. (12) yields

$$\begin{aligned} & \frac{\partial N_{rr}}{\partial r} + \frac{1}{r} \frac{\partial N_{r\theta}}{\partial \theta} + \frac{1}{r} (N_{rr} - N_{\theta\theta}) + I_0 r \Omega^2 \\ & = I_0 \frac{\partial^2 u_0}{\partial t^2} + I_1 \frac{\partial^2 \varphi_r}{\partial t^2} - 2I_0 \Omega \frac{\partial v_0}{\partial t} - 2I_1 \Omega \frac{\partial \varphi_\theta}{\partial t}, \end{aligned} \tag{18}$$

$$\begin{aligned} & \frac{\partial N_{r\theta}}{\partial r} + \frac{1}{r} \frac{\partial N_{\theta\theta}}{\partial \theta} + \frac{2}{r} N_{r\theta} = I_0 \frac{\partial^2 v_0}{\partial t^2} \\ & + I_1 \frac{\partial^2 \varphi_\theta}{\partial t^2} + 2I_0 \Omega \frac{\partial u_0}{\partial t} + 2I_1 \Omega \frac{\partial \varphi_r}{\partial t}, \end{aligned} \tag{19}$$

$$\begin{aligned} & \frac{\partial Q_{rr}}{\partial r} + \frac{1}{r} \frac{\partial Q_{\theta\theta}}{\partial \theta} + \frac{Q_{rr}}{r} + \frac{\partial N_{rr}}{\partial r} \frac{\partial w_0}{\partial r} + N_{rr} \frac{\partial^2 w_0}{\partial r^2} \\ & + \frac{N_{\theta\theta}}{r^2} \frac{\partial^2 w_0}{\partial \theta^2} + \frac{2N_{r\theta}}{r} \frac{\partial^2 w_0}{\partial r \partial \theta} + \frac{1}{r} \frac{\partial N_{r\theta}}{\partial r} \frac{\partial w_0}{\partial \theta} \\ & + \frac{N_{rr}}{r} \frac{\partial w_0}{\partial r} + \frac{1}{r^2} \frac{\partial N_{\theta\theta}}{\partial \theta} \frac{\partial w_0}{\partial \theta} + \frac{1}{r} \frac{\partial N_{r\theta}}{\partial \theta} \frac{\partial w_0}{\partial r} \\ & - c_t \frac{\partial w_0}{\partial t} + F_i(t) \delta(w_s - w) = I_0 \left(\frac{\partial^2 w_0}{\partial t^2} - \Omega^2 \frac{\partial^2 w_0}{\partial \theta^2} \right), \end{aligned} \tag{20}$$

$$\begin{aligned} & \frac{\partial M_{rr}}{\partial r} + \frac{1}{r} \frac{\partial M_{r\theta}}{\partial \theta} + \frac{1}{r} (M_{rr} - M_{\theta\theta}) - Q_{rr} + I_1 r \Omega^2 \\ & = I_1 \frac{\partial^2 u_0}{\partial t^2} + I_2 \frac{\partial^2 \varphi_r}{\partial t^2} - 2I_1 \Omega \frac{\partial v_0}{\partial t} - 2I_2 \Omega \frac{\partial \varphi_\theta}{\partial t}, \end{aligned} \tag{21}$$

$$\begin{aligned} & \frac{\partial M_{r\theta}}{\partial r} + \frac{1}{r} \frac{\partial M_{\theta\theta}}{\partial \theta} + \frac{2}{r} M_{r\theta} - Q_{\theta\theta} = \\ & I_1 \frac{\partial^2 v_0}{\partial t^2} + I_2 \frac{\partial^2 \varphi_\theta}{\partial t^2} + 2I_1 \Omega \frac{\partial u_0}{\partial t} + 2I_2 \Omega \frac{\partial \varphi_r}{\partial t}. \end{aligned} \tag{22}$$

Herein, $N_{rr}, N_{\theta\theta}, N_{r\theta}, M_{rr}, M_{\theta\theta}, M_{r\theta}, Q_{rz}, Q_{\theta z}$ are defined as follows

$$\begin{aligned} & \begin{bmatrix} N_{rr} \\ N_{\theta\theta} \\ N_{r\theta} \end{bmatrix} = \int_{-\frac{h}{2}}^{\frac{h}{2}} \begin{bmatrix} \sigma_{rr} \\ \sigma_{\theta\theta} \\ \sigma_{r\theta} \end{bmatrix} dz, \quad \begin{bmatrix} M_{rr} \\ M_{\theta\theta} \\ M_{r\theta} \end{bmatrix} = \int_{-\frac{h}{2}}^{\frac{h}{2}} z \cdot \begin{bmatrix} \sigma_{rr} \\ \sigma_{\theta\theta} \\ \sigma_{r\theta} \end{bmatrix} dz, \\ & \begin{bmatrix} Q_{rz} \\ Q_{\theta z} \end{bmatrix} = \int_{-\frac{h}{2}}^{\frac{h}{2}} K_s \begin{bmatrix} \sigma_{rz} \\ \sigma_{\theta z} \end{bmatrix} dz, \end{aligned} \tag{23}$$

And $\{I_0, I_1, I_2\} = \int_{-h/2}^{h/2} \rho(z) \{1, z, z^2\} dz,$

3. Solution method

Closed-form solutions for nonlinear transient responses of rotating circular plates are derived herein via Galerkin discretization. Through assumed mode decomposition of displacement fields, expansions satisfying simply supported ends are formulated

$$\begin{aligned}
 [u_0, \varphi_r] &= \sum_{m=1}^{\infty} \sum_{n=1}^{\infty} [U(t), \Phi_r(t)] \cos\left(\frac{\pi mr}{R}\right) \sin(n\theta), \\
 [v_0, \varphi_\theta] &= \sum_{m=1}^{\infty} \sum_{n=1}^{\infty} [V(t), \Phi_\theta(t)] \sin\left(\frac{\pi mr}{R}\right) \cos(n\theta), \\
 w_0 &= \sum_{m=1}^{\infty} \sum_{n=1}^{\infty} W(t) \sin\left(\frac{\pi mr}{R}\right) \sin(n\theta),
 \end{aligned} \tag{24}$$

The parameters m and n correspond to the wave numbers in the radial and azimuthal directions, with both indices spanning the positive integer domain (1, 2, ...). Substituting Eq. (24) into Eqs. (18)-(22), one can obtain

$$\sum_{m=1}^M \sum_{n=1}^N \left(\mathbf{I} \cdot \begin{bmatrix} \frac{\partial^2 U(t)}{\partial t^2} \\ \frac{\partial^2 V(t)}{\partial t^2} \\ \frac{\partial^2 W(t)}{\partial t^2} \\ \frac{\partial^2 \Phi_r(t)}{\partial t^2} \\ \frac{\partial^2 \Phi_\theta(t)}{\partial t^2} \end{bmatrix} + \mathbf{S} \cdot \begin{bmatrix} U(t) \\ V(t) \\ W(t) \\ \Phi_r(t) \\ \Phi_\theta(t) \\ W(t)^2 \\ W(t)^3 \\ U(t) \cdot W(t) \\ V(t) \cdot W(t) \\ \Phi_r(t) \cdot W(t) \\ \Phi_\theta(t) \cdot W(t) \end{bmatrix} + \mathbf{\Xi} \cdot \begin{bmatrix} \frac{\partial U(t)}{\partial t} \\ \frac{\partial V(t)}{\partial t} \\ \frac{\partial W(t)}{\partial t} \\ \frac{\partial \Phi_r(t)}{\partial t} \\ \frac{\partial \Phi_\theta(t)}{\partial t} \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ F_c(t) \delta(w_i - w) \\ 0 \\ 0 \end{bmatrix} \right) = 0 \tag{25}$$

The Galerkin discretization isolates transverse displacement inertia within the resulting ordinary differential equations.

$$\sum_{m=1}^M \sum_{n=1}^N \left(\mathbf{M} \cdot \begin{bmatrix} \frac{\partial^2 U(t)}{\partial t^2} \\ \frac{\partial^2 V(t)}{\partial t^2} \\ \frac{\partial^2 W(t)}{\partial t^2} \\ \frac{\partial^2 \Phi_r(t)}{\partial t^2} \\ \frac{\partial^2 \Phi_\theta(t)}{\partial t^2} \end{bmatrix} + \mathbf{K} \cdot \begin{bmatrix} U(t) \\ V(t) \\ W(t) \\ \Phi_r(t) \\ \Phi_\theta(t) \\ W(t)^2 \\ W(t)^3 \\ U(t) \cdot W(t) \\ V(t) \cdot W(t) \\ \Phi_r(t) \cdot W(t) \\ \Phi_\theta(t) \cdot W(t) \end{bmatrix} + \mathbf{C} \cdot \begin{bmatrix} 0 \\ 0 \\ \frac{\partial W(t)}{\partial t} \\ 0 \\ 0 \end{bmatrix} + \mathbf{F} \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \end{bmatrix} \right) = 0 \tag{26}$$

Here, $[\mathbf{M}]$, $[\mathbf{K}]$, and $[\mathbf{C}]$ respectively represent the mass, stiffness, and damping matrices, and $\mathbf{F} = \iint_A F_c(t) \delta(w_s - w) \cdot \sin\left(\frac{m\pi}{\ell} r\right) \sin(n\theta) dA$. In the end, Runge-Kutta method is employed to solve Eq. (26).

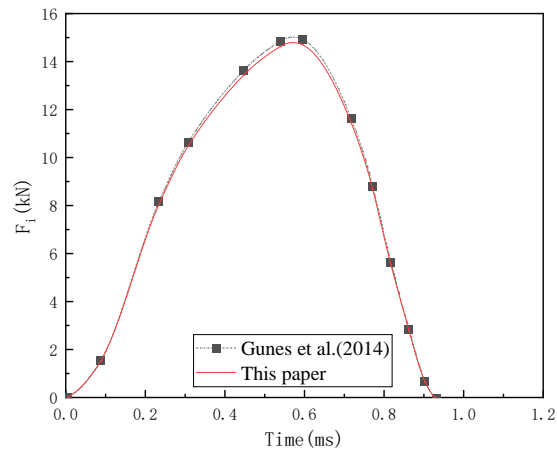


Fig. 2 Comparative analysis of contact force time histories ($\Omega=0$): Present findings vs. Gunes *et al.* (2014)

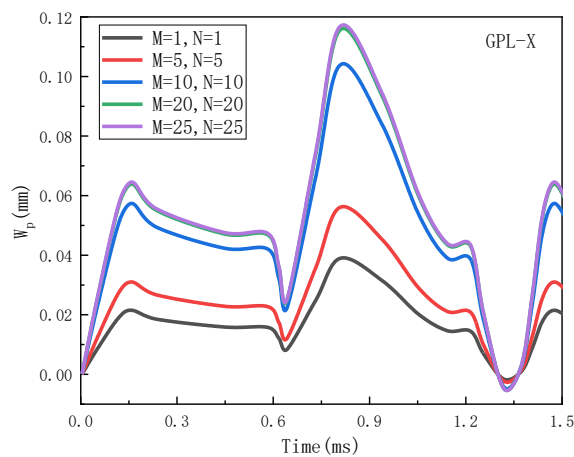


Fig. 3 Convergence verification (GPL-X, Porosity-III, $e_1=0.3$, $w_{GPL}=0.1\%$, $\Omega=200$ rad/s)

4. Numerical analysis

Due to the fact that the spinning circular plates under low-velocity impact remain unstudied. For validation, while the low-velocity impact behavior of FG-GPL porous structures remains unexplored, this study validates its methodology by replicating the functionally graded circular plate impact test from Gunes *et al.* (2014). Key parameters—adapted in our MATLAB code—include plate properties and the impactor are the same as Gunes *et al.* (2014), as depicted in Fig. 2, our FSDT-based results align closely with Gunes’ 3D elasticity solution, exhibiting a maximum discrepancy of <2%.

In the following part, during the calculation: For the circular plate: Radius $R=0.75$ m, $h=0.1$ m, $E_m=130$ GPa, $\nu_m=0.34$, $\rho_m=8960$ kg/m³, For the GPL reinforcement: $E_{GPL}=1.01$ TPa, $\rho_{GPL}=1062.5$ kg/m³, $\nu_{GPL}=0.186$, $w_{GPL}=1.5$ μ m, $l_{GPL}=2.5$ μ m, $t_{GPL}=15$ nm; For the spherical impactor (ball): $R_s=0.05$ m, $E_s=210$ GPa, $\rho_s=7850$ kg/m³, $\nu_s=0.3$; Impact velocity $V=5$ m/s.

Numerical reliability necessitates convergence verification. Fig. 3 demonstrates time-history

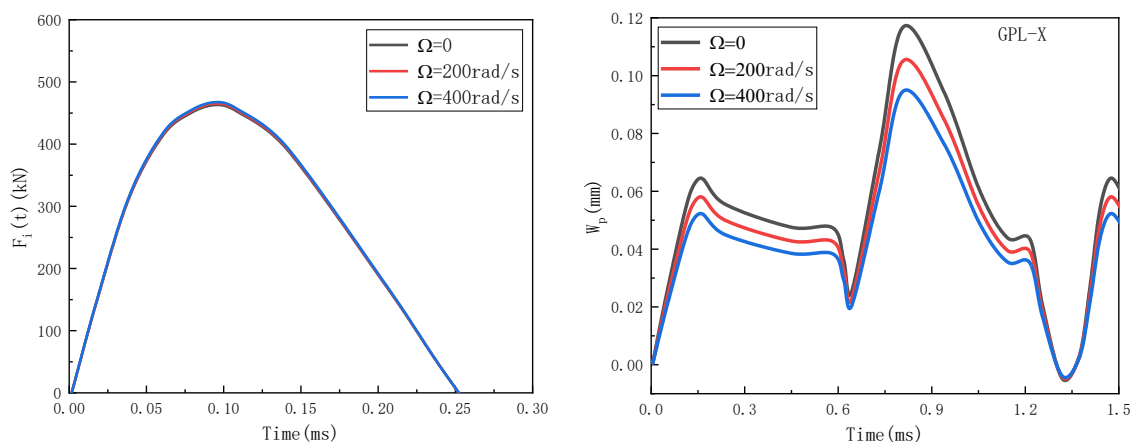


Fig. 4 Effect of spinning velocity (GPL-X, Porosity-III, $e_1=0.3$, $w_{GPL}=0.1\%$)

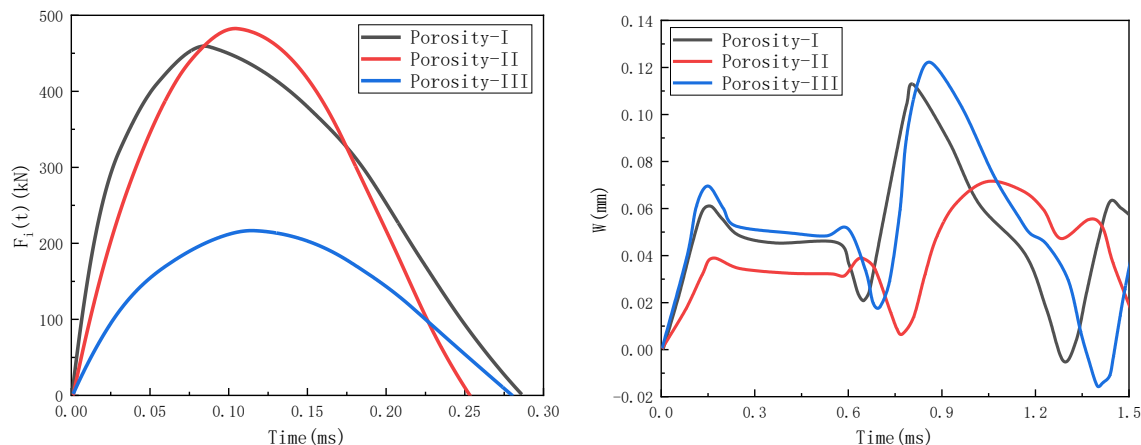


Fig. 5 Effect of porosity distribution (GPL-X, $e_1=0.2$, $w_{GPL}=0.1\%$, $\Omega=200$ rad/s)

convergence for the circular plate's nonlinear transient dynamics under uniform blast load. Solutions stabilize at $M=N=20$, subsequently adopted as the computational standard.

Fig. 4 systematically investigates the influence of spinning velocity (Ω) on contact force and displacement dynamics. Quantitative analysis reveals two markedly distinct trends: contact force magnitude exhibits negligible variation (fluctuation $< 2.7\%$ within $\Omega=0-400$ rad/s), while contact displacement demonstrates significant rotational speed dependence, decreasing from 0.11734 mm to 0.0948 mm as rotational speed increases from 0 to 400 rad/s, with maximum displacement occurring at $t=0.095$ ms when the plate reaches peak deformation. Mechanistic interpretation: this divergence originates from rotational hardening effects. Centrifugally induced circumferential stress fields constrain structural deformation while exerting minimal influence on local contact mechanics governed by graphene dispersion morphology. The stabilizing centrifugal potential suppresses displacement-amplifying vibration modes, yet remains subordinate to contact stiffness within load transmission pathways.

Fig. 5 reveals distinct patterns in contact behavior across different porosity distributions. Porosity-II demonstrates the highest contact force among the tested configurations, while Porosity-

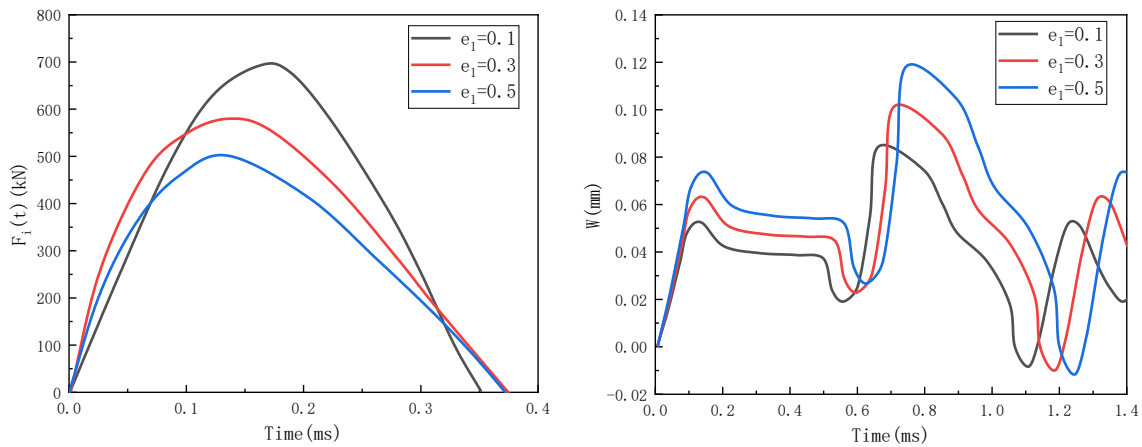


Fig. 6 Effect of porosity coefficient (GPL-X, Porosity-I, $w_{GPL}=0.1\%$, $\Omega=200$ rad/s)

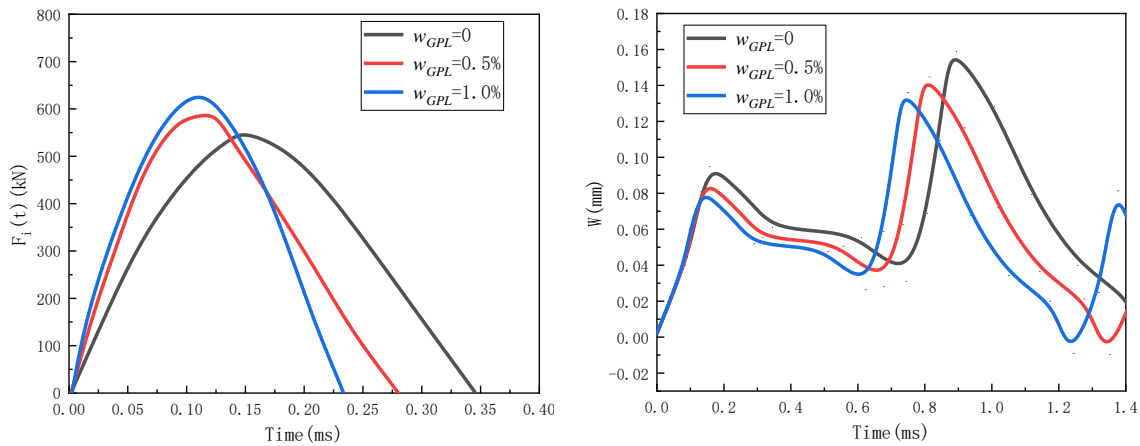


Fig. 7 Effect of porosity coefficient (GPL-X, Porosity-I, $e_1=0.5$, $\Omega=200$ rad/s)

III shows the lowest contact force magnitude. Conversely, Porosity-II exhibits the smallest contact displacement, with Porosity-III presenting the most significant displacement response. These observations suggest an inverse relationship between porosity concentration and mechanical resistance, where increased porosity correlates with reduced load-bearing capacity and greater deformation under applied forces.

Fig. 6 shows that the porosity coefficient e_1 significantly affects the contact behavior between materials. As e_1 increases, the contact force generally decreases due to reduced effective contact area and load-bearing capacity. Meanwhile, the contact displacement also shows a decreasing trend with higher e_1 values, as the porous structure provides more compliance and energy absorption capacity. This inverse relationship occurs because greater porosity creates more internal voids and weaker structural integrity, leading to lower resistance to deformation. The combined effects demonstrate how material porosity influences mechanical interactions, where both contact force and displacement diminish with increasing e_1 , reflecting the compromised structural stiffness and altered deformation characteristics of porous materials.

Fig. 7 depicts that the mass fraction w_{GPL} of graphene platelets plays a crucial role in

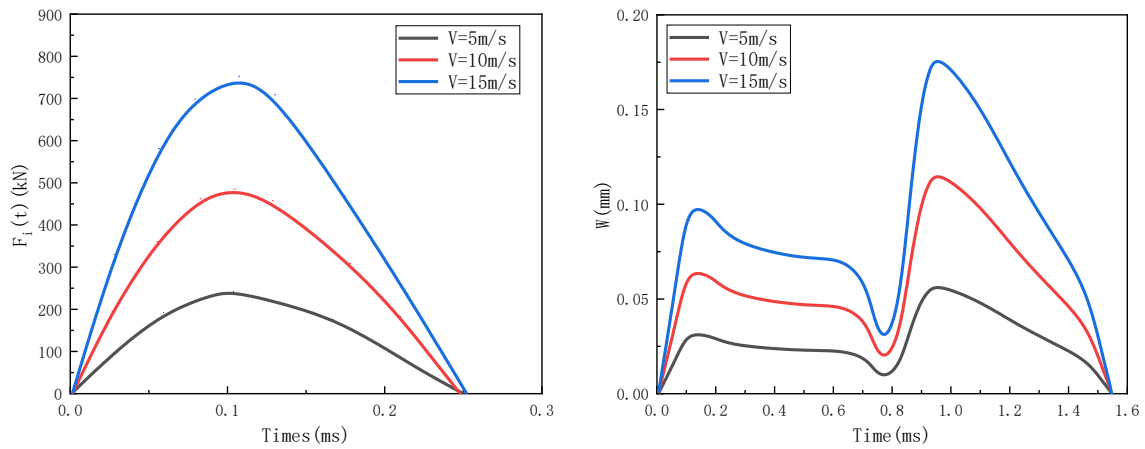


Fig. 8 Effect of impact velocity V (GPL-X, Porosity-I, $e_1=0.2$, $w_{GPL}=0.1\%$, $\Omega=200$ rad/s)

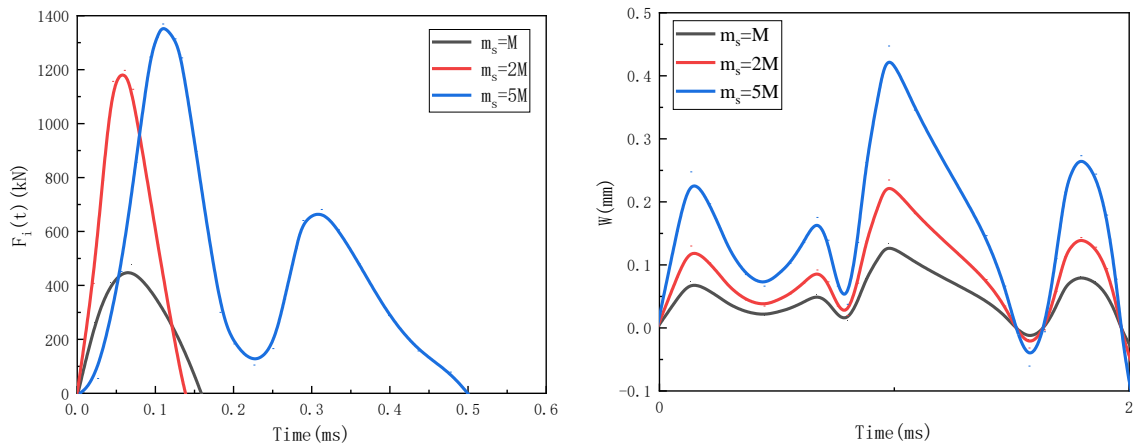


Fig. 9 Effect of impactor mass m_s (GPL-X, Porosity-III, $e_1=0.2$, $w_{GPL}=0.1\%$, $\Omega=200$ rad/s)

determining the contact behavior of composite materials. As w_{GPL} increases, the contact force generally rises due to enhanced stiffness and load-bearing capacity from the reinforcing graphene platelets. Conversely, the contact displacement tends to decrease with higher w_{GPL} , as the improved structural integrity restricts deformation. This dual effect arises because graphene reinforcement strengthens the material matrix, increasing resistance to external forces while reducing flexibility. The observed trends highlight how graphene content modulates mechanical interactions, where greater w_{GPL} leads to higher contact forces but smaller displacements, reflecting the trade-off between reinforcement and deformation characteristics.

Fig. 8 shows that the impact velocity V significantly influences the dynamic contact behavior between colliding bodies. As V increases, both the contact force and displacement exhibit a rising trend. Higher impact velocities generate greater kinetic energy, leading to intensified force transmission at the contact interface. Simultaneously, the enhanced energy input causes more pronounced material deformation, resulting in larger contact displacements. This correlated growth occurs because faster impacts induce stronger stress waves and inertial effects, which collectively amplify the mechanical response. The observed pattern demonstrates how impact velocity governs

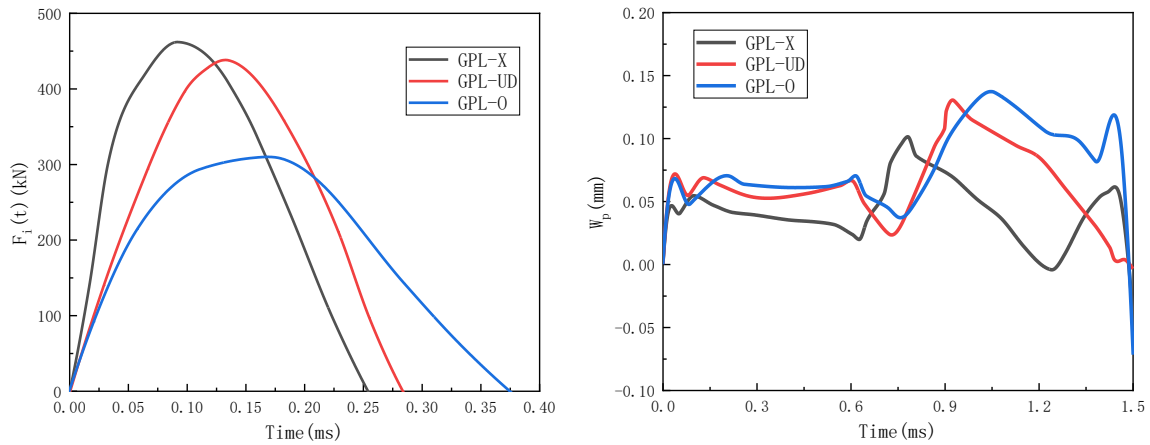


Fig. 10 Effect of GPLs distributions (Porosity-III, $e_1=0.3$, $w_{GPL}=0.1\%$, $\Omega=200$ rad/s)

collision dynamics, where elevated V values simultaneously escalate contact forces and displacements, reflecting the energy-dependent nature of impact interactions.

Fig. 9 shows the influence of impactor mass (m_s) on contact behavior reveals complex dynamic interactions. As m_s increases from M to $5M$, the contact force exhibits progressive intensification with a notable transition: while $m_s=M$ or $2M$ produces single-peak force profiles, the $m_s=5M$ case develops dual-peak characteristics. This bifurcation likely arises from stress wave reflections within the impactor and target material, where higher mass enables sequential energy release phases. Concurrently, contact displacement amplifies with growing m_s due to enhanced momentum conservation and prolonged contact duration. The dual-peak phenomenon specifically suggests inertial dominance at critical mass ratios, where initial impact energy partitions into primary compression and secondary rebound phases. Larger displacements correlate with deeper material indentation, as increased kinetic energy at higher m_s overcomes structural resistance more effectively. These observations highlight mass-dependent energy redistribution mechanisms, where wave propagation timing and inertial delays jointly sculpt the force-displacement relationship, particularly manifesting in the distinctive dual-peak signature at $m_s=5M$.

Fig. 10 shows that under the impact of the striker ball, the GPL-X distribution pattern exhibits the highest contact force magnitude, while the GPL-O distribution pattern yields the lowest contact force magnitude. Conversely, the GPL-X pattern demonstrates the smallest impact-induced displacement, whereas the GPL-O pattern results in the largest contact displacement.

5. Conclusions

This study investigates the dynamic contact behavior of nanocomposite plates under coupled physical fields, focusing on four key variables: graphene platelet (GPL) distribution patterns, rotational velocity, porosity architectures, and impactor mass. The research reveals that GPL spatial configuration fundamentally alters material stiffness characteristics, while rotational effects induce selective hardening that modifies global displacement without affecting local contact forces. Porosity distribution is shown to create distinct tradeoffs between mechanical resistance and energy absorption. Most notably, a critical mass threshold triggers a dynamic regime shift in

contact behavior, manifested through dual-peak force profiles resulting from stress wave interference phenomena. These findings collectively demonstrate how multi-scale material design and external loading conditions synergistically govern impact resistance mechanisms.

- (1) GPL distribution patterns determine load transfer efficiency, with X-type configurations exhibiting superior stiffness enhancement compared to O-type arrangements.
- (2) Rotational hardening provides a unique mechanism for controlling structural displacement independently of contact force characteristics.
- (3) Porosity architectures create predictable performance tradeoffs between mechanical resistance and energy dissipation capacity.
- (4) Impactor mass governs contact dynamics through inertial effects, with higher masses inducing sequential energy release phases.
- (5) The emergence of dual-peak force profiles beyond a critical mass ratio indicates stress wave dominance in energy partitioning.

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